

CUSTOMIZATION	AIRCRAFT TYPES	DOCTYPES	REVISION DATE	REVISION NUMBER	TITLE
DLH	A318 A321 A320 A319	AMM	01-Nov-2022	129	73-13-42-720-010-A02 - Functional Test of the Fuel Recirculation Cooling System without Barfield Tester and with AIDS
TAIL NUMBER - MSN - FSN: D-AIDQ - 05028 - 364					
Print date: 2023-01-16 03:33:38					

**** ON A/C FSN 101-149, 151-153, 155-199, 351-399**

TASK 73-13-42-720-010-A02

Functional Test of the Fuel Recirculation Cooling System without Barfield Tester and with AIDS

WARNING: PUT THE SAFETY DEVICES AND THE WARNING NOTICES IN POSITION BEFORE YOU START A TASK ON OR NEAR:

- THE FLIGHT CONTROLS
- THE FLIGHT CONTROL SURFACES
- THE LANDING GEAR AND THE RELATED DOORS
- COMPONENTS THAT MOVE.

MOVEMENT OF COMPONENTS CAN KILL OR CAUSE INJURY TO PERSONS AND/OR CAN CAUSE DAMAGE TO THE EQUIPMENT.

WARNING: MAKE SURE THAT THE GROUND SAFETY-LOCKS ARE IN POSITION ON THE LANDING GEAR. THIS WILL PREVENT UNWANTED MOVEMENT OF THE LANDING GEAR, AND THUS POSSIBLE INJURY TO PERSONS AND DAMAGE TO THE AIRCRAFT AND/OR EQUIPMENT.

1. Reason for the Job

Self explanatory

2. Job Set-up Information

A. Fixtures, Tools, Test and Support Equipment

REFERENCE	QTY	DESIGNATION
No specific	AR	SAFETY BARRIER(S)
No specific	AR	WARNING NOTICE(S)
98D27803500001	1	LOCKING TOOL ZERO POS

B. Work Zones and Access Panels

ZONE/ACCESS	ZONE DESCRIPTION
211	COCKPIT
400	POWER PLANT, NACELLES AND PYLONS

C. Referenced Information

REFERENCE	DESIGNATION
Ref. 12-11-28-650-003-A	Pressure Refuel with Automatic Control
Ref. 12-11-28-650-004-A	Pressure Refuel with Manual Control
Ref. 27-50-00-866-008-A	Extension of the Flaps on the Ground
Ref. 27-50-00-866-008-A01	Extension of the Flaps/Slats on the Ground
Ref. 31-36-00-740-008-A	Access to the Parameter Call-Up Menus
Ref. 31-60-00-860-001-A	EIS Start Procedure
Ref. 71-00-00-710-012-B	Test No.3 : Idle Leak Check
Ref. TSM 73-22-00-810-843	Failure of the Heat Management System with Operation of the Mode 5 on Engine 1
Ref. TSM 73-22-00-810-844	Failure of the Heat Management System with Operation of the Mode 5 on Engine 2

3. Job Set-up

SUBTASK 73-13-42-941-054-A

A. Safety Precautions

- (1) Make sure that the flap/slat control lever on panel 114VU is in the zero position [Ref. AMM TASK 27-50-00-866-008](#).
- (2) Make sure that the LOCKING TOOL ZERO POS ([98D27803500001](#)) is on the slat/flap control lever.
- (3) On the center pedestal, on the ENG panel 115VU:
 - (a) Put a WARNING NOTICE(S) to tell persons not to start the engine.
- (4) Put a WARNING NOTICE(S) on the throttle control lever to tell persons not to operate the reverse system.
- (5) Put the SAFETY BARRIER(S) in position.

SUBTASK 73-13-42-865-056-A

CUSTOMIZATION	AIRCRAFT TYPES	DOCTYPES	REVISION DATE	REVISION NUMBER	TITLE
DLH	A318 A321 A320 A319	AMM	01-Nov-2022	129	73-13-42-720-010-A02 - Functional Test of the Fuel Recirculation Cooling System without Barfield Tester and with AIDS
TAIL NUMBER - MSN - FSN: D-AIDQ - 05028 - 364					
Print date: 2023-01-16 03:33:38					

B. Make sure that this(these) circuit breaker(s) is(are) closed:

PANEL	DESIGNATION	FIN	LOCATION
49VU	ENGINE/2/FADEC A/AND EIU 2	2KS2	A05
49VU	ENGINE/1/FADEC A/AND EIU 1	2KS1	A04
121VU	ENGINE/ENG2/FADEC B	4KS2	Q40
121VU	ENGINE/ENG1/FADEC B/AND EIU 1	4KS1	R41
121VU	FUEL/PUMPS/2/R CTL AND L IND	10QA	R29
121VU	FUEL/CTR TK/L&R XFR/CTL & IND	2QL	R26
121VU	FUEL/CTR TK/L&R XFR/CTL & IND	1QL	R25
121VU	FUEL/PUMPS/1/L CTL AND R IND	3QA	R23

SUBTASK 73-13-42-010-055-A
DELETED

4. Procedure

**** ON A/C FSN 157-199, 351-399**

SUBTASK 73-13-42-720-053-C

A. Functional Test of the Fuel Recirculation Cooling System

(1) Make sure that circuit breakers 1QL and 2QL are closed.

NOTE: If the related DC BUS is not powered and the circuit breakers not closed, the applicable Fuel Level Sensing Control Unit (FLSCU) cannot supply and IDG/FRV open command via the Engine Interface Unit (EIU) to the Electronic Engine Control (EEC).

- (2) Do the Electronic Instrument System (EIS) start procedure (ECAM only) [Ref. AMM TASK 31-60-00-860-001](#).
- (3) On the ECAM control panel, push the FUEL pushbutton switch.
- (4) On the ECAM lower Display Unit (DU), make sure that the FUEL page is shown.
- (5) On the lower ECAM DU, monitor the FUEL system display and make sure that:
 - all the fuel pumps are shown green in-line
 - the L(R) inner wing temperature is less than 52.5 deg.C
 - the L(R) outer cell temperature is less than 55 deg.C
 - the L(R) wing tank contains more than 290 kg (639.34 lb) of fuel to make sure that the IDG shut off sensor 38QJ1(2) is wet. Refuel if necessary [Ref. AMM TASK 12-11-28-650-003](#) or [Ref. AMM TASK 12-11-28-650-004](#)
 - the cross-feed valve is open.
- (6) Do the functional test of the recirculation cooling system:

(a) Do the idle leak check [Ref. AMM TASK 71-00-00-710-012](#):

1 Before you start the test, make sure that the engine oil temperature is 30 deg.C (86.00 deg.F). You can read this value on the lower ECAM DU.

NOTE: Fuel recirculation is not possible if engine oil temperature is less than 30 deg.C (86.00 deg.F).

2 On the upper ECAM DU make sure that the fuel flow is less than 4410 lbs/h and the fuel temperature is more than 5 deg.C (41.00 deg.F).

NOTE: Fuel recirculation is not possible if fuel flow is more than 4410 lbs/h and/or fuel temperature is less than 5 deg.C (41.00 deg.F).

3 Do this test:

a Get access to the parameter label call-up menu through the MCDU [Ref. AMM TASK 31-36-00-740-008](#).

ACTION	RESULT
1.On the LABEL CALL-UP page, set with the MCDU keyboard: 07C/1/245/01 for engine 1. 07C/2/245/10 for engine 2.	
2.Push the line key 1L.	07C, 1, 245, 01 (07C, 2, 245, 10) come into view in their related box with the label 245 bits.

4 Decode the fuel return value position as follows.
© AIRBUS S.A.S. ALL RIGHTS RESERVED. CONFIDENTIAL AND PROPRIETARY DOCUMENT.

CUSTOMIZATION	AIRCRAFT TYPES	DOCTYPES	REVISION DATE	REVISION NUMBER	TITLE
DLH	A318 A321 A320 A319	AMM	01-Nov-2022	129	73-13-42-720-010-A02 - Functional Test of the Fuel Recirculation Cooling System without Barfield Tester and with AIDS
TAIL NUMBER - MSN - FSN: D-AIDQ - 05028 - 364					
Print date: 2023-01-16 03:33:38					

a Read the label 245 bits 28 to 14.

b Refer to the following table to find the value of each significant bit (set to 1 bit).

!Bit number!	28	27	26	25	24	23	22	21	20	19	18	17	16	15	14
!value of	!	!	!	!	!	!	!	!	!	!	!	!	!	!	!
!the bit	!	!	!	!	!	!	!	!	!	!	!	!	!	!	!
!when set	!16384!	8192!	4096!	2048!	1024!	512!	256!	128!	64!	32!	16!	8!	4!	2!	1!
!to 1	!	!	!	!	!	!	!	!	!	!	!	!	!	!	!
	!	!	!	!	!	!	!	!	!	!	!	!	!	!	!

c Add all the significant bit values and calculate the position of the fuel return valve (in percent) with the following equation:

$$\text{Valve position} = \frac{\text{addition of the value of the significant bits}}{32768} \times 128$$

- Example: date transmitted on bits 14 to 28 of label 245

00011111010000
28-----14

.Value of the significant bits = $2048+1024+512+256+128+64+16 = 4048$

$$\text{.Therefore valve position} = \frac{4048}{32768} \times 128 = 15.8 \text{ percent.}$$

5 Make sure that the calculated valve position is more than 10 percent.

(7) If the above procedure is not satisfactory, continue the troubleshooting using [Ref. TSM 73-22-00-810-843](#) or [Ref. TSM 73-22-00-810-844](#) to further isolate the reason for the test failure.

**** ON A/C FSN 101-149, 151-153, 155-199, 351-399**

5. Close-up

SUBTASK 73-13-42-410-054-A
DELETED

SUBTASK 73-13-42-942-055-A

A. Removal of Equipment

- (1) Make sure that the work area is clean and clear of tools and other items.
- (2) Remove the SAFETY BARRIER(S).
- (3) Remove the LOCKING TOOL ZERO POS ([98D27803500001](#)) from the flap/slat lever on panel 114VU.
- (4) Remove the WARNING NOTICE(S).
- (5) Remove the access platform(s).
- (6) Remove the ground support and maintenance equipment, the special and standard tools and all other items.